

Process design of Absorber

The gases entering the absorber per day are :

Carbon disulfide, CS_2 = 459.92 Kmols

Hydrogen disulfide, H_2S = 1314.06 Kmols

Methane, CH_4 = 73.0 Kmols

Total Kmols of gases entering = 1846.98 Kmols

Gases leaving the absorber from the top are :

Hydrogen disulfide, H_2S = 1307.49 Kmols

Methane, CH_4 = 73.0 Kmols

Carbon disulfide, CS_2 = 2.30 Kmols

For liquid phase,

At top of tower,

X_t = mols of carbon disulfide/mols of oil

At bottom of tower,

X_b = mols of carbon disulfide/mols of oil

For Gas phase,

At top of tower,

Y_t = mols of carbon disulfide / mols of inert gases

At bottom of tower,

Y_b = mols of carbon disulfide / mols of inert gases

We take pure oil at the inlet

hence,

$X_t = 0$

We have to find X_b

$Y_t = 2.3/(1307.49 + 73.0)$

$$= 0.001666$$

$$Y_b = 459.92 / (1314.06 + 73.0)$$

$$= 0.33158$$

Taking the heat of solution of carbon disulfide in hydrocarbon oil as negligible, the equilibrium curve is plotted by :

$$X : 0.00989 \quad 0.0909 \quad 0.2307 \quad 0.333 \quad 0.412$$

$$Y : 0.0052 \quad 0.0538 \quad 0.222 \quad 0.475 \quad 0.866$$

From the equilibrium curve, $X_b^* = 0.280$

$$(L_m'/G_m')_{\min} = (Y_b - Y_t) / (X_b^* - X_b) = 1.20$$

Taking (L_m' / G_m') as 1.2 times that of $(L_m'/G_m')_{\min}$

Hence, $(L_m' / G_m') = 1.44$

$$L_m' = 1.44 \times G_m'$$

$$G_m' = 1387.06 \text{ Kmols}$$

$$\text{Hence, } L_m' = 1.44 \times 1387.06 = 1997.32 \text{ Kmols}$$

And $X_b = 0.235$

$$G_{mb} = G_m' (1 + Y_b)$$

$$= 1387.06 \times (1 + 0.33157)$$

$$= 1846.96 \text{ Kmols}$$

$$L_{mb} = L_m' (1 + X_b)$$

$$= 1997.3 \times (1 + 0.235)$$

$$= 2466.74 \text{ Kmols}$$

$$G_{mt} = G_m' (1 + Y_t)$$

$$= 1387.06 \times (1 + 0.001666)$$

$$= 1389.37 \text{ Kmols}$$

$$L_{mt} = L_m'(1 + X_t)$$

$$= 1997.36 \text{ Kmols}$$

Average molecular weights :

$$(M_G)_{Top} = (73.0/1382.8) \times 16.0 + (1307.49/1382.8) \times 34.05 \\ + (2.30/1382.8) \times 76.1$$

$$= 33.167 \text{ Kg/Kmol}$$

$$(M_G)_{Bottom} = (459.92/1846.98) \times 76.1 + (1314.06/1846.98) \times 34.05 \\ + (73.0/1846.98) \times 16.0$$

$$= 43.80 \text{ Kg/Kmol}$$

Density of gases = PM_{avg}/RT

The column is operating at 40°C and 1 atm

Density of gases :

$$(\rho_g)_{top} = (PM_{avg}/RT)_{top} = [(1.013 \times 10^5 \times 33.8167)/(8314 \times 313)]$$

$$= 1.2911 \text{ kg /m}^3$$

$$(\rho_g)_{bottom} = (PM_{avg}/RT)_{bottom} = [(1.013 \times 10^5 \times 43.80)/(8314 \times 313)]$$

$$= 1.705 \text{ kg /m}^3$$

$$G_b = 1846.96 \times 43.80$$

$$= 80,896.84 \text{ Kgs}$$

$$G_t = 1389.31 \times 33.167$$

$$= 46081.23 \text{ Kgs}$$

$$L_b = 2466.96 \times 160.26$$

$$= 3,95,319.75 \text{ Kgs}$$

$$L_t = 1997.36 \times 180.0 = 3,59,524.80 \text{ Kgs}$$

$$[(L_b / G_b) (\rho_g / \rho_l)^{1/2}]_{\text{top}} = 7.8019 / (1.2911 / 810)^{1/2}$$

$$= 0.3115$$

$$[(L_b / G_b) (\rho_g / \rho_l)^{1/2}]_{\text{bottom}} = 4.8867 / (1.705 / 160.26)^{1/2}$$

$$= 0.2242$$

Since, the ratio is more at the top of the tower, we will consider the flooding conditions at the top of the tower.

For 80% flooding conditions and $[(L_b / G_b) (\rho_g / \rho_l)^{1/2}] = 0.3115$

The value of $(G_f^2 F_p \psi \mu^{0.2}) / (\rho_g \rho_l g) = 0.044$

Where, $\psi = (\rho_{\text{water}} / \rho_l) = 1.2345$

μ is viscosity of liquid = 1.7 cP

Choosing 75mm Ceramic Intalox saddles as packing material.

Packing Factor, $F_p = 70 \text{ m}^{-1}$

Approximate surface area, $a = 92 \text{ (m}^2/\text{m}^3)$

Percent void space, $\varepsilon = 80\%$

Nominal Size, $d_p = 75 \text{ mm}$

Hence, $G_f = 2.1674 \text{ Kg/m}^2\text{-s}$

And, $G = 0.8 \times G_f = 1.73393 \text{ Kg/m}^2\text{-s}$

Cross sectional area of the absorber, A_c :

$$A_c = G_b / G = 0.30759 \text{ m}^2$$

Hence, the diameter of the column, D_c :

$$D_c = 0.630 \text{ m}$$

Check : $D_c / d_p = 0.63 / 0.075 = 8.4$,

This is less than the required minimum ratio of 10.

This packing is not suitable for the column.

Choosing 25 mm Ceramic Intalox saddles as packing material.

Packing Factor, $F_p = 320 \text{ m}^{-1}$

Approximate surface area, $a = 255 \text{ (m}^2/\text{m}^3)$

Percent void space, $\varepsilon = 77\%$

Nominal Size, $d_p = 25 \text{ mm}$

We get $G_f = 1.013169 \text{ Kg/m}^2\text{-s}$

$G = 0.81095 \text{ Kg/m}^2\text{-s}$

$A_c = 0.65768 \text{ m}^2$

$D_c = 0.9150 \text{ m}$

Check : $D_c/d_p = 36.60$, which is greater than the required minimum ratio of 10.

Hence, the packing material is suitable.

Wetting rate calculation :

Wetting Rate = $L_w = L'/(A_c \rho_l a')$

$L_w = [359524.8/(3600 \times 24)]/[0.65768 \times 810 \times 255]$

$= 3.063 \times 10^{-5} \text{ m}^3/\text{m-s}$

$= 1.18696 \text{ ft}^3/\text{ft-hr}$

As, this wetting rate is greater than the required minimum of $0.85 \text{ ft}^3/\text{ft-hr}$ the wetting rate is acceptable.

Pressure Drop calculation :

$\Delta P = C_2 \times 10^{(C_3 U_{tl})} \times \rho_g \times U_{tg}^2$

Where, u_{tl} : superficial liquid velocity

And, U_{tg} : superficial gas velocity

$L = 6.327 \text{ Kg/m}^2\text{-s} = 4655.40 \text{ Lb/ft}^2\text{-hr}$

For this value of L,

And for saddles of 1" : $c_2 = 0.31$, $c_3 = 0.0222$

Also, the range of L : 2,520 – 14,440 Lb/ft²-hr

The calculated L lies safely in this range.

$G = 0.810953 \text{ Kg/m}^2\text{-s} = 596.707 \text{ Lb/ft}^2\text{-hr}$

Hence, the superficial liquid and gas velocities :

$U_{tl} = 7.811 \times 10 \text{ m/s} = 0.029744 \text{ m/s}$

$U_{tg} = 0.6281 \text{ m/s} = 2.607 \text{ ft/s}$

$\Delta P = 0.0678 \text{ in water/ft packing}$

$= 5.6499 \text{ mm water/m packing}$

Tower Height Calculations :

$$Z = H_{OG} \times N_{OG}$$

Where,

Z : height of tower

H_{OG} : height of overall gas transfer unit

N_{OG} : number of overall gas transfer unit

$$H_{OG} = H_G + (m G_m / L_m) H_L$$

Cornell's Method :

The generalized equation of Cornell et al, for gas phase transfer units, H_G

$$H_G = (0.029 \psi D^{1.11} Z^{0.33} Sc_g^{0.5}) / (L f_1 f_2 f_3)^{0.5}$$

ψ obtained from graph for 80% flooding conditions = 50 m

$$Sc_g = \mu_g / \rho_g D_g$$

$$f_1 = (\mu_l / \mu_w) = 1.00886$$

$$f_2 = (\rho_l / \rho_w) = 1.30134$$

$$f_3 = (\text{surface tension of the liquid} / \text{surface tension of water}) = 0.9379$$

D_g : diffusion coefficient for mixture of gases

For multi component gas phase diffusivity,

Blanc's law :

$$D_{i,mix} = 1 / [\sum (x_j / D_{ij})]$$

$$D_{AB} = [10^3 T^{1.75} \{ (M_A + M_B) / (M_A M_B) \}^{1/2} / [P \{ (\sum V_A)^{1/3} + (\sum V_B)^{1/3} \}^2]$$

$$D_{CS_2-H_2S} = 0.1153 \text{ cm}^2/\text{s}$$

$$D_{CS_2-CH_4} = 0.1471 \text{ cm}^2/\text{s}$$

$$\text{Hence, } D_{g \text{ mix}} = 0.1169 \text{ cm}^2/\text{s}$$

$$S_{C_g} = 0.89445$$

$$H_G = [(0.029)(50)0.915^{1.11} Z^{0.33} 0.98445^{0.5}] / [6.327(1.00886)(1.301348)(0.9379)]^{0.5}$$

$$H_G = 0.44518 X Z^{0.33}$$

The generalized equation of Cornell et al, for liquid phase transfer units, H_L

$$H_L = (\Phi C / 3.28) (\mu_l / \rho_l D_l)^{1/2} (Z / 3.05)^{0.15}$$

Where, C : correction factor

Φ : correlation parameter = 0.06, for $L = 4655.4 \text{ Lb/ft}^2\text{-hr}$

μ_L = liquid viscosity , Pa.s

ρ_L = Liquid density , kg/m^3

D_L = liquid –diffusion coefficient, m^2/s

Z =height of packing,m

Wilke Chang equation :

$$D_{AB} = [7.4 \times 10^{-8} (\phi M_B)^{1/2} T] / \mu_l V_A$$

$$\mu_l = 1.7 \text{ cP}$$

$$D_L = 1.534 X 10^5 \text{ cm}^2/\text{s}$$

$$H_L = 0.10858 X Z^{0.15}$$

$$H_{OG} = 0.44518 Z^{0.33} + 0.10858 Z^{0.15} \text{ mG}_m/L_m$$

N_{OG} Calculations :

$$N_{OG} = \int_{Y_t}^{Y_b} [dY / (Y - Y^*)] - [\frac{1}{2} \ln \{(1 + Y_b) / (1 + Y_t)\}]$$

Y	Y*	1/(Y - Y*)
0.05	0.0425	133.3
0.10	0.085	66.6
0.15	0.125	40.0
0.20	0.1675	30.7
0.25	0.21	25.0
0.30	0.25	20.0
0.33157	0.2775	18.5
0.35	0.290	16.7

Plotting a graph of $1/(Y - Y^*)$ Vs Y,

$$\text{Area under the curve from } Y_t \text{ to } Y_b = \int_{Y_t}^{Y_b} [dY / (Y - Y^*)] = 9.5$$

$$\text{And the value of } [\ln\{(1 + Y_b)(1 + Y_t)\}/2] = 0.142$$

$$N_{OG} = 9.5 - 0.142 = 9.358$$

$$(G_m / L_m)_t = 1389.37/1997.36 = 0.6956$$

$$(mG_m / L_m) = 0.83472$$

$$(G_m / L_m)_b = 0.75486$$

$$(mG_m / L_m)_b = 0.9058$$

$$(mG_m / L_m)_{avg} = 0.87028$$

$$Z = H_{OG} N_{OG}$$

$$= [H_G + (m G_m / L_m) H_L] N_{OG}$$

$$Z = 4.16599 Z^{0.33} + 0.88428 Z^{0.15}$$

By trial, Z is calculated to be,

$$Z = 10$$

Hence the height of the tower, Z = 10 mts

Mechanical Design of Absorber

Material used for making the shell is carbon steel.

Thickness of shell, t_s :

$$t_s = [(p D) / (2f J - p)] + c$$

Where,

Inner Diameter of vessel, $D_i = 0.9150$ m

Working Pressure = 1.013×10^5 N/m²

Design Pressure, $p = 1.05 \times 1.013 \times 10^5 = 1.063 \times 10^5$ N/m²

Permissible Stress, $f = 95 \times 10^6$ N/m²

J= Joint Efficiency = 0.85

Corrosion allowance, $c = 2$ mm

Hence, $t_s = 2.6$ mm

taking thickness as 8 mm

Thus, outer diameter of shell $D_o = 0.915$ m + 2×0.008 m = 0.931 m

Axial Stress Due to Pressure :

Axial stress due to pressure, f_{ap}

$$f_{ap} = [(p D_i) / \{4(t_s - c)\}]$$
$$= 4.052 \times 10^6 \text{ N/m}^2$$

Stress due to Dead Load :

a) Compressive Stress due to weight of shell up to a distance x

Density of Shell material = $\rho_s = 7700$ kg /m³

$$f_{ds} = [\pi/4 (D_o^2 - D_i^2) \rho_s x] / \pi /4 (D_o^2 - D_i^2)$$
$$= 7.7 \times 10^4 \text{ X N/m}^2$$

b) Compressive stress due to weight of insulation at height x :

Insulator used is asbestos

Thickness of insulation $t_{ins} = 3$ cm

Diameter of insulation, D_{ins}

Density of insulation = 2200 kg / m³

Mean diameter of vessel = D_m

For large diameter column

$$D_{ins} = D_m$$

$$f_{dins} = [\pi D_{ins} t_{ins} \rho_{ins} x] / [\pi D_m (t_s - c)] \\ = 0.11 \times 10^6 \times N / m^2$$

c) Compressive stress due to liquid in column up to height x :

Density of liquid, $\rho_l = 810 \text{ kg/m}^3$

$$f_{dliq} = [(\pi / 4) D_i^2 \times \rho_l \times x] / [\pi D_m (t_s - c)] \\ = 30.3 \times 10^3 \text{ N/m}^2$$

d) Compressive stress due to packing and attachments :

- 1) Packing weight
- 2) Head weight
- 3) Ladder

Density of packing (Ceramic Intalox saddles) = 705 kg/m^3

$$1) \text{ Packing Weight} = (\pi / 4) D_i^2 \times x \times \text{density of packing} \times 9.81 \\ = 4547.68 \times N$$

$$2) \text{ Head weight} = [(\pi / 4) D_i^2 \times x \times 810 \times 9.81] = 5225 \times N$$

$$3) \text{ Weight of Ladder} = 1600 \times N$$

Total compressive stress due to attachments f_d is given by

$$f_{d(\text{attachments})} = (\text{Packing Weight} + \text{Head weight} + \text{Ladder}) / [\pi D_i (t_s - c)] \\ = (263674.22 + 395713.1) \times N$$

Stress due to Wind :

Stress due to wind is given by

$$f_{wx} = M_w / Z$$

Where,

$$M_w = [(0.7 \rho_w D_o x^2) / 2]$$

$$Z = [(\pi / 4) D_o^2 (t_s - c)]$$

$$\text{Pressure due to wind} = \rho_w = 0.05 \times v_w^2$$

Considering velocity of wind to be 150 km/hr

$$p_w = 1125 \text{ N/m}^2$$

$$f_{wx} = \{1.4 p_w X^2\} / \{\pi D_o (t_s - c)\}$$

$$= 89.75 \times 10^3 x^2 \text{ N}$$

Determining the value of x :

$$f_{tmax} = 95 \times 10^6 \text{ N/m}^2$$

$$f_{tmax} = f_{wx} + f_{ap} - f_{dx}$$

$$95 \times 10^6 (0.85) = 89.75 \times 10^3 x^2 + 4.052 \times 10^6 - [7.7 \times 10^4 x + 0.11 \times 10^6 x + 30.3 \times 10^3 x + 263674.22 x + 395713.1 x]$$

$$\text{Gives, } x = 32.1 \text{ m}$$

As this value is greater than the design tower height, hence the shell thickness can withstand the tensile and compressive stresses.

Design of Gasket and Bolt Size calculations :

$$\text{Width of gasket} = N = 10 \text{ mm}$$

Gasket material is asbestos

$$\text{Gasket factor} = m = 2$$

$$\text{Minimum design seating stress} = Y_a = 11.2 \text{ N/mm}^2$$

$$\text{Basic gasket seating width } b_o = N/2$$

$$b_o = 10/2 = 5 \text{ mm}$$

Effective gasket seating width

$$b = 2.5 (b_o)^{1/2} = 5.6 \text{ mm}$$

$$\text{Flange inner diameter} = D_{fi} = 0.915 \text{ m}$$

$$\text{Flange outer diameter} = D_{fo} = 0.931 \text{ m}$$

$$\text{Mean diameter, } G = (D_{fi} + D_{fo})/2$$

$$= 0.923 \text{ m}$$

Under atmospheric conditions, the bolt load due to gasket is given by

$$W_{m1} = \pi b G Y_a$$

$$= 0.16238 \times 10^6 \text{ N}$$

Design pressure = p

$$= 1.064 \times 10^5 \text{ N/m}^2$$

After the internal pressure is applied, the gasket which is compressed earlier, is released to some extent and the bolt load is given by :

$$W_{m2} = \pi(2b) G \times m \times p + (\pi/4) G^2 p$$

$$= 0.07868 \times 10^6 \text{ N}$$

Bolt used is hot rolled carbon steel

f_a is permissible tensile stress in bolts under atmospheric condition

f_b is permissible tensile stress in bolts under operating condition

$$f_a = 58.7 \times 10^6 \text{ N/m}^2$$

$$f_b = 54.5 \times 10^6 \text{ N/m}^2$$

A_m is the area of bolt

$$A_{m1} = W_{m1}/f_a$$

$$A_{m2} = W_{m2}/f_b$$

$$A_{m1} = 0.00276 \text{ m}^2$$

$$A_{m2} = 0.00144 \text{ m}^2$$

Number of bolts = mean diameter / b_o x 2.5

$$= 66 \text{ bolts}$$

To determine the size of bolts, the larger of above two areas should be considered

$$\text{Diameter of bolts} = [(A_{m1} / \text{Number of bolts}) \times (4/\pi)]^{1/2}$$

$$= 0.729 \text{ cm}$$

FLANGE THICKNESS :

Thickness of flange = t_f

$$t_f = [G\sqrt{(p/K f)}] + c$$

Where,

$$K = 1/[0.3 + (1.5 W_m h_G)/(H \times G)]$$

Hydrostatic end force = $H = (\pi / 4) G^2 p$

$$= 71192.60 \text{ N}$$

h_G is radial distance from gasket load reaction to bolt circle,

$$h_G = (B - G)/2$$

$$= 0.0133$$

$$B = \text{outside diameter of gasket} + 2 \times \text{diameter of bolt} + 12\text{mm} \\ = 0.949 \text{ m}$$

$$W_m = 0.16238 \times 10^6 \text{ N}$$

$$K = 2.8628$$

Hence the thickness of flange = 20.25 mm

HEAD DESIGN: FLANGED & SHALLOW :

Material stainless steel

$$\text{Permissible stress, } f = 130 \text{ N/mm}^2$$

$$\text{Design pressure, } p = 1.064 \times 10^5 \text{ N/m}^2$$

Stress identification factor W is given by

$$W = \left(\frac{1}{4}\right) [3 + (R_c/R_1)^{1/2}]$$

$$\text{Crown Radius, } R_c = 0.915 \text{ m}$$

$$\text{Knuckle radius, } R_1 = 0.055 \text{ m}$$

$$\text{Stress identification factor, } W = 1.77$$

$$\text{Thickness of head} = t_h = (p \times R_c \times W)/(2f)$$

$$t_h = 0.91 \text{ mm}$$

So, we can take thickness of head as that of thickness of shell.

NOZZLE THICKNESS :

Material Carbon steel

$$\text{Considering diameter of nozzle, } D_n = 0.25 \text{ m}$$

$$\text{Thickness of nozzle} = t_n$$

Material is Stainless steel.

$$\text{Permissible stress} = 130 \times 10^6 \text{ N/m}^2$$

$$J = 0.85$$

$$t_n = (p D_n) / (2 f J - p)$$

$$t_n = 0.12 \text{ mm}$$

No corrosion allowance, since the material is stainless steel.

$$\text{Thickness} = 3\text{mm}$$

SUPPORT FOR ABSORBER :

Material used is structural steel (IS 800)

Skirt support is used.

Inner Diameter of the vessel, $D_i = 0.915$ m

Outer Diameter of the vessel, $D_o = 0.931$

Height of the vessel, $H = 10$ m

Density of carbon steel, $\rho_s = 7700$ kg /m³

Density of oil, $\rho_l = 810$ kg /m³

Total weight = Weight of vessel + Weight of Attachments (liquid + packing + head
+ ladder)

$$= [(\pi/4) (D_o^2 - D_i^2) \times H \rho_s \times 9.81 + (\pi /4) D_i^2 \times H \times \rho_l \times 0.77 + (\pi /4) D_i^2 \times H \times \rho_p \\ + 52250 + 1600 \times H]$$

$$= 77.174 \times 10^3 \text{ N}$$

Diameter of Skirt is 0.915 m

Considering the height of Skirt is 2 m

Wind Pressure is 1125 N/m²

Thickness of the skirt support is t_{sk} .

Stress due to dead load :

$$f_d = \text{Total Weight} / \pi D_s t_{sk} \\ = 26.847 \times 10^3 / t_{sk} \text{ N/m}^2$$

Stress due to wind load :

The forces due to wind load are determined as :

$$p_w = k p_1 h_1 D_o$$

Where K is coefficient depending on the shape factor.

$K = 0.7$ for cylindrical surface

$$p_1 = 89.75 \times 10^3 \times^2 \\ = 8.975 \times 10^6 \text{ N/m}^2$$

$$p_w = k p_1 h_1 D_o \\ = 59.49 \times 10^6 \text{ N/m}^2$$

Bending moment due to wind at the base of the vessel is determined by

$$M_w = p_{iw} \times (h_1/2) \\ = 292.45 \times 10^6 \text{ N-m}$$

$$f_{wb} = (4 \times M_w) / (\pi D_o t_{sk}) \\ = 39.99 \times 10^6 / t_{sk}$$

Stress due to Seismic Load

$$\text{Load} = C W$$

$$W \text{ is total Weight of vessel} = 77.174 \times 10^3 \text{ N}$$

$$C \text{ is Seismic Coefficient} = 0.08$$

$$f_{sb} = (2/3)[C W H / (\pi R_{ok}^2 t_{sk})]$$

where,

R_{ok} is radius of skirt

$$f_{sb} = 60.4617 \times 10^3 / t_{sk} \text{ N/m}^2$$

Maximum Stress at bottom of Skirt :

$$f_{tmax} = (f_{wb} \text{ Or } f_{sb}) - f_{db} \\ = (1.121 \times 10^6 / t_{sk}) \text{ N/m}^2$$

Permissible tensile Stress for structural steel = 140 N/mm²

$$t_{sk} = 0.008 \text{ m}$$

Hence thickness of skirt is 8 mm

Design of Condenser

The condenser is working with the refining column

Feed gas : $\text{CS}_2 = 657.03 \text{ Kmols} = 2083.33 \text{ Kg/hr}$

Entering at 80°C , Pressure is 1 atm.

Flow rate, $m = 0.5787 \text{ Kg/s}$

$Q' = m \lambda = 195.92 \text{ KW}$

Using water as coolant, flow rate of water required,

$m_w = 4.687 \text{ Kg/s}$

for a counter current operation, LMTD:

$\Delta T_{\text{In}} = 44.81^\circ\text{C}$

Calculation of heat transfer area :

Assuming heat transfer coefficient, $U = 500 \text{ W/m}^2\text{-K}$

Dirt factor = $0.003 \text{ ft}^2\text{h-K/Btu}$

Heat transfer Area, A :

$A = 8.744 \text{ m}^2$

Choosing $\frac{3}{4}$ " OD, 8 ft long tubes,

H.T.A = 0.14593 m^2

$N_t = 60$ tubes

Choosing a 1-2 Heat exchanger, TEMA P or S type $15/16$ " triangular pitch,

Nearest tube count = 62 tubes

Shell I.D = 254 mm

Corrected heat transfer area, $A = 9.008 \text{ m}^2$

Corrected $U = 533.874 \text{ W/m}^2\text{-K}$

Fluid velocities :

a) Tube side : Number of passes, $N_p = 2$

Flow area, $a_t = 6.04 \times 10^{-3} \text{ m}^2$

Tube side velocity, $v_t = 0.85219 \text{ m/s}$

For the tube side the acceptable velocity range is 1-2 m/s. Hence, this velocity is unacceptable.

Trial 2 :

Choosing TEMA P or S type for 1 shell and 4 tube passes, 1" triangular pitch.

$N_t = 62$

Shell I.D = 305 mm

$v_t = 1.704 \text{ m/s}$, which is within the acceptable limits.

$a_t = 3.02 \times 10^{-3}$

b) Shell side :

D_s : Shell I.D

Cross flow area at the center of the shell, S_m :

$$S_m = [(P' - D_o) L_s] D_s / P'$$
$$= 0.01159 \text{ m}^2$$

Taking 25% baffle cut, $L_c = 76.25 \text{ mm}$

$L_s = 0.5 D_i = 152.5 \text{ mm}$

$N_b + 1 = 16$

$N_b = 15$ baffles

Summary of shell and tube details :

Number of tube passes	: 4
Number of shell passes	: 1
Tube I.D	: 15.75 mm
Tube O.D	: 19.05 mm
Shell I.D	: 305 mm
Number of tubes	: 62
Tube pitch	: 1" = 25.4 mm
Length of tube	: 2.348 m
Baffle pitch	: 152 mm
Number of baffles	: 15
P_p	: 21.99 mm
L_c	: 76.25 mm
$U_{corrected}$: 533.874 W/m ² -K

Film Transfer Coefficients :

a) Shell side (condensing vapors) :

Cooling water is heated from 30°C to 40°C

Wall temperature, $T_w = (1/2)[80 + (30+40)/2] = 57.5^\circ\text{C}$

Film temperature, $T_f = (80+57.5)/2 = 68.75^\circ\text{C}$

Using properties of CS₂ at this temperature,

Thermal conductivity, $K = 0.15427 \text{ W/m-K}$

Density, $\rho = 1225 \text{ Kg/m}^3$

Viscosity, $\mu = 0.28 \text{ cP}$

Reynold's number (N_{Re}) = 209.41

$$h = 1.51 [(K^3 \rho^2 g) / \mu]^{1/3} \text{Re}^{-1/3}$$
$$= 2246.33 \text{ W/m}^2\text{-K}$$

b) Tube side (water) :

Flow area, $a_t = 3.02 \times 10^{-3} \text{ m}^2$

Density, $\rho = 1000 \text{ Kg/m}^3$

Viscosity, $\mu = 0.85 \text{ cP}$

Prandtl 's number (N_{Pr}) = 5.995

$$h_i d_i / K = 0.023 (N_{Re})^{0.8} (N_{Pr})^{0.3}$$
$$= 156.51$$

hence, $h_i = 4876.86 \text{ W/m}^2\text{-K}$

$$\text{wall resistance} = (x_w/k_w) \times [d_o/(d_i - d_o)] \times \ln(d_o/d_i)$$
$$= 4.02646 \times 10^{-5}$$

Overall Heat Transfer Coefficient :

$$U_{od} = 792.55 \text{ W/m}^2\text{-K}$$

Pressure Drop calculations :

a) Tube side (water) :

$$F = 0.079 (N_{Re})^{-1/4} = 5.926 \times 10^{-3}$$

$$\Delta P_L = [(4 F v_t^2)/(2g d_i)] \rho g$$
$$= 5309.53 \text{ N/ m}^2$$

$$\Delta P_E = 2.5(\rho_t v_t^2/2)$$
$$= 3629.52 \text{ N/ m}^2$$

$$(\Delta P_T)_{\text{total}} = N_p (\Delta P_L + \Delta P_E)$$
$$= 35.756 \text{ Kpa}$$

This value is less than the maximum value of allowable pressure drop of 70 Kpa
Hence, it is acceptable.

b) Shell side :

Bell's method:

$$(N_{Re}) = 209.41 ,$$
$$\text{Friction factor, } f_K = 0.4$$

(i) Pressure drop in cross flow sections,

There are $(N_b - 1)$ crosses.

$$\Delta P_c = \left[\frac{(b f_K w^2 N_c)}{(\rho_f S_m^2)} \right] / (\mu_w / \mu_b)^{0.14} \text{ KN/m}^2$$

$$b = 2 \times 10^{-3}$$

$$N_c = D_s [1 - 2(L_c/D_s)] / P_p = 7$$

$$S_m = [(P' - D_o)L_s] D_s / P' = 0.011628 \text{ m}^2$$

$$\Delta P_c = 0.0132 \text{ KPa}$$

(ii) Pressure drop in the end zones,

There are two end zones - entrance and exit.

$$\Delta P_e = \Delta P_c [1 + N_{cw}/N_c] \text{ KN/m}^2$$

Number of effective cross flow zones in each window, N_{cw} :

$$N_{cw} = 0.8 L_c / P_p = 3$$

$$\text{Hence, } \Delta P_e = 0.01885 \text{ KN/ m}^2$$

(iii) Pressure drop in the window zones,

There are N_b zones

$$\Delta P_w = \left[\frac{b w^2 (2 + 0.6 N_{cw})}{(S_m S_w \rho_f)} \right] \text{ KN/m}^2$$

$$b = 5 \times 10^4$$

S_w = area for flow in the window zones (m^2)

$$S_w = S_{wg} - S_{wt}$$

Where, S_{wg} = gross window area (m^2)

S_{wt} = area occupied by tubes

$$S_{wg} = 23 \text{ sq. in} = 0.01483 \text{ m}^2, \text{ for } L_c/D_s = 0.25$$

$$S_{wt} = (N_T/8) (1 - F_c) \times 3.14 \times D_o^2 \text{ (m}^2)$$

$$= 3.04 \times 10^{-3} \text{ m}^2$$

hence, $S_w = 0.01179 \text{ m}^2$

$$\Delta P_w = 0.003788 \text{ KPa}$$

Hence, the total pressure drop,

$$\begin{aligned} (\Delta P_s)_{\text{total}} &= 2\Delta P_e + (N_b - 1) \Delta P_c + N_b \Delta P_w \\ &= 0.2963 \text{ Kpa} \end{aligned}$$

As, this pressure drop is lesser than the maximum allowable pressure drop in the shell side of 70 Kpa. It is acceptable.

MECHANICAL DESIGN OF CONDENSER:

SHELL SIDE:

Material : Carbon steel
No. of shells : 1
No. of passes : 4
Fluid : carbon disulphide
Internal diameter : 305 mm
Working pressure : 0.1 N/mm^2
design pressure : 0.11 N/mm^2
Inlet temperature : 80°C
Outlet temperature: 80°C
Allowable stress : 950 Kg/Cm^2

TUBE SIDE:

Material : Stainless steel (IS grade 10)
No. of tubes : 62
Outside diameter : 19.05 mm
Length : 2.438 m
fluid : water
pitch : 25.44 mm (triangular)
Allowable stress : 10.06 Kg/m^2
working pressure : 1.033 Kg/cm^2
Design pressure : 1.55 Kg/cm^2
Inlet temperature : 30°C
Outlet temoerature : 40°C

SHELL SIDE

SHELL THICKNESS:

$$\begin{aligned}t_s &= (P_d \times D_s)/(2fJ-P_d) \\ &= (0.11 \times 305)/((2 \times 95 \times 0.85) - 0.11) \\ &= 0.20 \text{ mm}\end{aligned}$$

But minimum thickness of shell is 6 mm

Therefore with corrosion allowance of 2mm

Thickness of shell = 8 mm

NOZZLE DIAMETER

$$\begin{aligned}M &= \text{Mass velocity/sec} \\ &= 2083.33/3600 \\ &= 0.5787 \text{ Kg/sec}\end{aligned}$$

Density $\rho = 1225 \text{ kg/m}^3$

Assume velocity to be 1 m/sec

$$\begin{aligned}(\pi \times d_n^2 \times \rho \times v)/4 &= M \\ d_n^2 &= (0.5787 \times 4)/(1 \times 3.14 \times 1225) \\ d_n &= 0.0245 \text{ m}\end{aligned}$$

NOZZLE THICKNESS

$$\begin{aligned}t_n &= (P_d \times d_n)/(2fJ-P) \\ &= (0.11 \times 24.5)/(2 \times 95 \times 0.85 - 0.11) \\ &= 0.016 \text{ mm}\end{aligned}$$

Nozzle thickness with corrosion allowance = 5 mm

HEAD THICKNESS:

$$\begin{aligned}t_h &= (P_d \times R_c \times W)/(2fJ) \\ W &= (1/4)(3 + (R_c/R_K)^{1/2})\end{aligned}$$

R_c - crown radius 80% of shell I.D. = 244 mm

R_K - Knuckle radius 10% of shell I.D. = 30.5 mm

$$W = 1.46$$

$$t_h = 0.242 \text{ mm}$$

Using same thickness as that of the shell = 8 mm

BAFFLE ARRANGEMENT:

Tansverse baffles

Number of baffles = 15

Baffle Spacing = $D_s = 152$ mm

Thickness of baffles = 6 mm

Height of baffle = $0.75 \times D_s$
= 114 mm

TIE RODS AND SPACERS:

For shell diameter $D_s = 152$ mm

No. of tie rods =

Diameter of rods = 13 mm

FLANGE CALCULATION:

Flange material = IS:2004-1962 class 2

Bolting steel = 5% Cr Mo steel

Gasket material = asbestos composition

Shell inside diameter = 305 mm

Shell thickness $t_s = 8$ mm

Shell outside diameter = $(2 \times t_s) + 305$
= $(2 \times 8) + 305$
= 321 mm

Allowable stress of flange material = 100 MN/m^2

Allowable stress of bolting material = 138 MN/m^2

GASKET WIDTH:

$$G_o/G_i = [(y - P_d m) / (y - p_d (m + 1))]^{1/2}$$

m-Gasket factor = 2.75

y-Minimum design seating stress = 25.5 MN/m^2

$$G_o/G_i = [25.5 - (0.11 \times 2.75) / 25.5 - (0.11 \times 3.75)]^{1/2}$$
$$= 1.0022$$

Minimum gasket width N = 10 mm

Basic gasket seating width $b_o = N/2$

$$= 5 \text{ mm} < 6.3 \text{ mm}$$

Let, inside diameter of gasket = inside diameter of shell = 305 mm

$$G_i = 305 + (2 \times 8)$$

$$= 321 \text{ mm}$$

Mean gasket width = $G_i + N$

$$= 331 \text{ mm}$$

therefore $G = 331 \text{ mm}$

Estimation of bolt load:

Load due to design pressure $H = (\pi G^2 P_d) / 4$

$$= (\pi \times 0.331^2 \times 0.11) / 4$$

$$= 0.0094 \text{ MN}$$

Effective gasket sitting width $b = b_o = 5 \text{ mm}$ since $b < 6.3 \text{ mm}$

Load to keep joint tight under pressure $H_p = \pi(2b)GmP_d$

$$= 3.14 \times 0.01 \times 0.331 \times 2.75 \times 0.11$$

$$= 0.0031 \text{ MN}$$

Total operating load $W_o = H + H_p$

$$= 0.0094 + 0.0031$$

$$= 0.0125 \text{ MN}$$

Load to seat gasket under bolting up condition $W_b = \pi b G y$

$$= 3.14 \times 0.331 \times 0.005 \times 25.5$$

$$= 0.133 \text{ MN}$$

Since $W_b > W_o$, controlling load = 0.133 MN

Minimum bolting area:

Total cross sectional area of bolt under operating condition $A_{m1} = W_o / S_b$

S_b -nominal bolt stress at design temperature of $80^\circ\text{C} = 138 \text{ MN/m}^2$

Therefore $A_{m1} = 0.0125 / 138$

$$= 0.000091 \text{ m}^2$$

Total cross sectional area of bolt required for gasket seating $A_{m2} = W_b / S_a$

S_a -nominal bolt stress at ambient temperature (30°C) = 138 MN/m^2

Therefore $A_{m2} = 0.126 / 138$

$$=0.000913 \text{ m}^2$$

Since $A_{m2} > A_{m1}$, $A_m = A_{m2} = 0.000913 \text{ m}^2$

Calculation of optimum bolt size:

$$C = 2(R + g_1) + B$$

Choosing bolt M-18 × 2

$$\text{Total number of bolts} = G / (18 \times 2)$$

$$= 331 / 36$$

$$= 9$$

Actual number of bolts = 12

R-radial clearance from bolt circle to point of connection of hub and back of flange = 27 mm

B-inside diameter of flange = outside diameter of shell = 0.331 m

$$g_1 = g_o / 0.707, \text{ let } g_o = 8 \text{ mm}$$

$$g_1 = 0.008$$

$$C = 2(0.027 + 0.008) + 0.331$$

$$= 0.441 \text{ m}$$

Therefore bolt circle diameter = 0.441 m

Flange outside diameter:

$$A = C + \text{bolt diameter} + 0.02$$

$$= 0.441 + 0.018 + 0.02$$

$$= 0.479 \text{ m}$$

Check of gasket width:

$$A_b - \text{root area of bolt (m}^2\text{)} - 1.54 \times 10^{-4} \text{ m}^2$$

S_g -allowable stress for bolting material at atmospheric temperature = 138 MN/ m²

$$\text{Therefore, } nA_b S_g / \pi G N = 25.77$$

Since $25.77 < 2y$

Condition is satisfied.

Flange moment computations:

$$W_o = W_1 + W_2 + W_3 \text{ (under operating condition)}$$

$$W_1 = (\pi B^2 P_d) / 4$$

$$= (3.14 \times 0.321^2 \times 0.11) / 4$$

$$= 0.0090 \text{ MN}$$

$$W_2 = H - W_1$$

$$H = (\pi G^2 P_d) / 4$$

$$= (3.14 \times 0.331^2 \times 0.11) / 4$$

$$= 0.0095 \text{ MN}$$

$$W_2 = 0.0095 - 0.0090$$

$$= 0.0005 \text{ MN}$$

$$W_3 = W_o - H$$

$$= 0.0125 - 0.0095$$

$$= 0.003 \text{ MN}$$

Total flange moment,

$$M_o = W_1 a_1 + W_2 a_2 + W_3 a_3$$

$$a_1 = (C - B) / 2$$

$$= 0.032 \text{ m}$$

$$a_3 = (C - G) / 2$$

$$= 0.027 \text{ m}$$

$$a_2 = (a_1 + a_3) / 2$$

$$= 0.0295 \text{ m}$$

$$M_o = 0.0090 \times 0.032 + 0.0005 \times 0.0295 + 0.003 \times 0.027$$

$$= 3.838 \times 10^{-4} \text{ MNm}$$

Bolting up condition:

$$\text{Total flange moment } M_g = W a_3$$

$$W = (A_m + A_b) S_g / 2$$

$$= 0.0736 \text{ MN}$$

$$a_3 = 0.027 \text{ m}$$

$$\text{Therefore } M_g = 0.01953 \text{ MNm}$$

$$= 1.987 \times 10^{-3} \text{ MNm}$$

Since $M_g > M_o$ for moment under operating condition, M_g is controlling.

$$\text{Therefore } M = M_g = 1.987 \times 10^{-3} \text{ MNm}$$

Flange thickness:

$$t^2 = (M C_F Y) / B S_T$$

$$K = A / B$$

= Outer diameter of flange/ inner diameter of flange

$$= .479/0.331$$

$$= 1.447$$

$$Y = 14$$

Assume $C_F = 1$

Therefore thickness 't' = 0.055 m

Actual bolt spacing $B_S = \pi C/n$

$$= (3.14 \times 0.441)/12$$

$$= 0.115 \text{ m}$$

Bolt pitch correction factor $C_F = [B_S/(2d+t)]^{1/2}$

$$= [0.115/(2 \times 0.018 + 0.055)]^{1/2}$$

$$= 0.381$$

Therefore $C_F^{1/2} = 0.617$

Actual flange thickness = $C_F^{1/2} t$

$$= 0.617 \times 0.055$$

$$= 0.0339 \text{ m}$$

$$= 33.9 \text{ mm}$$

TUBE SIDE:

TUBE THICKNESS:

$$t_t = Pd_o/2fJ + P$$

J = 1 for seamless tube

$$\text{Therefore } t_t = (1.55 \times 19.05)/(2 \times 1006 + 1.55)$$

$$= 0.0146 \text{ mm}$$

No corrosion allowance since the tube is made of stainless steel

Thickness of tube = 1 mm

TUBE SHEET:

$$t_s = FG[0.25P/f]^{1/2}$$

F-The value of F varies according to type of heat exchanger, for most cases it is taken as 1

$$G = 331 \text{ mm}$$

$$\text{Therefore } t_s = 331[(0.25 \times 1.55)/1006]^{1/2}$$

$$= 6.49 \text{ mm}$$

CHANNEL AND CHANNEL COVER:

$$t = G[KP/f]^{1/2}$$

K = 0.3 for ring type gasket

Material of construction is carbon steel

So allowable stress $f = 950 \text{ Kg/cm}^2$

Therefore $t = 331[(0.3 \times 1.55)/950]^{1/2}$

$$= 7.323 \text{ mm}$$

With corrosion allowance $t = 10 \text{ mm}$

NOZZLE THICKNESS:

Assume inlet and outlet diameter = 75 mm

Thickness of nozzle $t_h = Pd/2fJ - P$

$$= (1.55 \times 75) / (2 \times 0.85 \times 950 - 1.55)$$

$$= 0.0612 \text{ mm}$$

with corrosion allowance thickness = 6mm

SADDLE SUPPORT DESIGN:

Material - Low carbon steel

Vessel diameter = 305 mm

Length of shell = 2.438 m

Torispherical head:

Crown radius = 244 mm

Knuckle radius = 30.5 mm

Working pressure = 1 Atm

Shell thickness = 8 mm

Head thickness = 8mm

Corrosion allowance = 2mm

permissible stress = 950 Kg/cm^2

R-Vessel radius = 152.5

Distance of saddle centre line from shell end $A = 0.45 \times R$

$$= 68.63 \text{ mm} < 0.2L$$

Longitudinal bending moment:

The bending moment at the support is;

$$M_1 = QA[1 - \{(1 - A/L) + (R^2 - H^2/2AL)/(1 + 4H/3L)\}]$$

$$A = 68.63 \text{ mm}$$

$$Q = W/2[L + 4H/3]$$

W-Weight of fluid and vessel.

Weight of shell material:

$$W_1 = [\pi(D_o^2 - D_i^2)L\rho_{\text{shell material}}] / 4$$

$$\rho_{\text{shell material}} = 7700 \text{ Kg/m}^3$$

$$W_1 = [3.14(0.331^2 - 0.305^2) \times 2.438 \times 7700] / 4$$

$$= 244 \text{ Kg}$$

Weight of tubes:

$$W_2 = n[\pi(D_o^2 - D_i^2)L\rho_{\text{tube material}}] / 4$$

$$\rho_{\text{tube material}} = 7800 \text{ Kg/m}^3$$

$$W_2 = 62[3.14(0.01905^2 - 0.01705^2) \times 2.438 \times 7800] / 4$$

$$= 66.8 \text{ Kg}$$

Weight of tube sheet:

$$W_3 = (2\pi D^2 t \rho) / 4$$

$$= (2 \times 3.14 \times 0.331^2 \times 0.0180 \times 7800) / 4$$

$$= 24.15 \text{ Kg}$$

Liquid load in the shell:

$$W_4 = (\text{shell volume} - \text{tube volume})\rho_{\text{liquid}}$$

$$= [(\pi D_s^2 L) / 4 - (n\pi d_o^2 l) / 4] \times 1175$$

$$= [(\pi \times 0.331^2 \times 2.438) / 4 - (62 \times \pi \times 0.01905^2 \times 2.438) / 4] \times 1175$$

$$= (0.2098 - 0.043) \times 1175$$

$$= 196 \text{ Kg}$$

Liquid load in tubes:

$$W_5 = n\pi d_i^2 l \rho_{\text{liquid}} / 4$$

$$= (62 \times 3.14 \times 0.01705^2 \times 2.438 \times 997) / 4$$

$$= 215.9 \text{ Kg}$$

Therefore total weight $W_T = W_1 + W_2 + W_3 + W_4 + W_5$
= 746 Kg

Hence, $Q = 746/2[2.438+(4 \times 0.257)/3]$
= 1037.19Kgm

$M_1 = (1037.19 \times 0.06863)[1 - \{(1 - 0.06863/2.44) + (0.1525^2 - 0.257^2)/2 \times 0.203 \times 4.88\} / (1 + 4 \times 0.257/3 \times 4.88)]$
= 2909.261[1 - {(0.958 + 0.0698)/1.0702}]
= 115.26 Kgm

The bending moment at the center of the span is given by:

$M_2 = (QL/4)[\{1 + 2(R^2 - H^2)/L^2\} / \{1 + (4H/3L)\} - (4A/L)]$
= $(1037.19 \times 2.438/4)[\{1 + 2(0.4525^2 - 0.257^2)/2.438^2\} / \{1 + (4 \times 0.257/3 \times 2.438)\} - (4 \times 0.203/2.438)]$
= 1234[1.312 - 0.1664]
= 1414 Kgm

Stress in shell at the saddle:

At the top most fiber of the cross-section,

$$f_1 = M_1 / (K_1 \pi R^2 t)$$

For an angle of 120° , $K_1 = 0.107$ m

t-thickness of shell = 8 mm

$$f_1 = 115.6 / (0.107 \times 3.14 \times 0.152^2 \times 0.008)$$
$$= 10.37 \text{ Kg/cm}^2$$

At the bottom most fiber of the cross-section

$$f_2 = M_1 / (K_2 \pi R^2 t)$$

For an angle of 120° , $K_2 = 0.192$ m

$$f_2 = 115.6 / (0.192 \times 3.14 \times 0.152^2 \times 0.008)$$
$$= 8.47 \text{ Kg/cm}^2$$

Stress in the shell at mid point:

$$f_3 = M_2 / (\pi R^2 t)$$

$$= 1414 / (3.14 \times 0.152^2 \times 0.008)$$
$$= 248 \text{ Kg/cm}^2$$

Thus the values of stresses are within the limited range

Hence the designed support is acceptable.